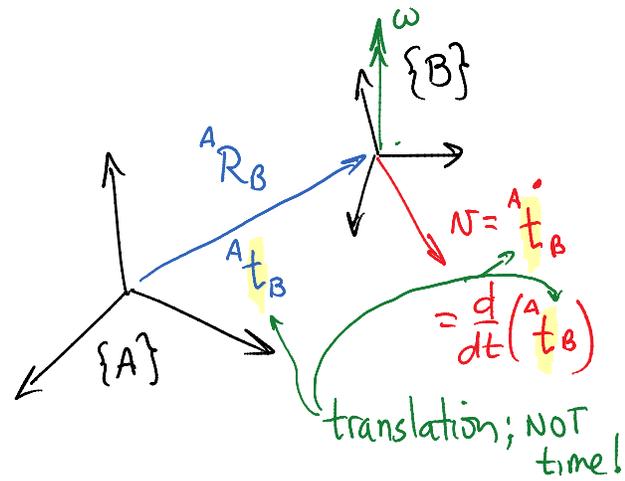


3.2_TimeVaryingCoordinateFrames

Wednesday, May 30, 2012
10:35 AM

At each instant of time a time-varying coordinate frame has specific translational and rotational velocity vectors.



N = translation velocity of a reference point (usually the origin of $\{B\}$).

$$N \in \mathbb{R}^3$$

$${}^A N_B = \frac{d}{dt} ({}^A t_B)$$

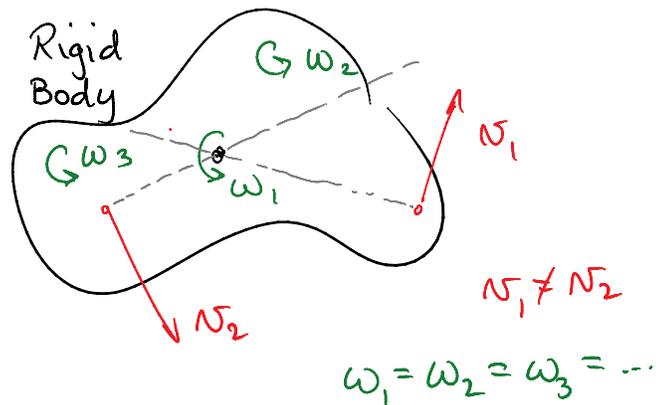
ω = rotational (or angular) velocity of a frame (or rigid body).

$$\omega \in \mathbb{R}^3$$

$${}^A \omega_B \neq \frac{d}{dt} ({}^A R_B)$$

$$\neq \frac{d}{dt} (\text{rpy})$$

$$\neq \frac{d}{dt} (\text{quaternion})$$



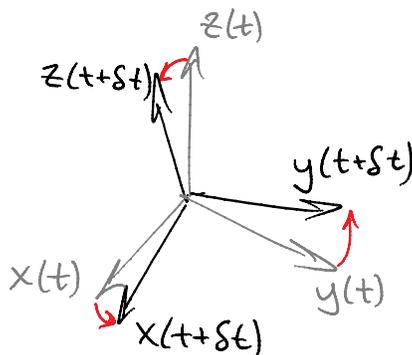
$\frac{\omega}{\|\omega\|} = \hat{\omega}$ defines an axis of rotation

$\|\omega\|$ defines the rate of rotation about the axis

3.2.1 Rotating Coordinate Frames

Goal: Define relationship for incremental changes in orientation and relate them to incremental changes in orientation parameters.

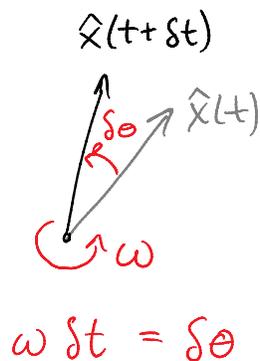
Time-varying frame \Rightarrow



Consider only the change in R . (Origin is fixed.)

A view of $\frac{d}{dt}(R(t)) = \dot{R}(t)$

$$\lim_{\delta t \rightarrow 0} \frac{\hat{x}(t+\delta t) - \hat{x}(t)}{\delta t} = \frac{d}{dt}(\hat{x}) = \dot{\hat{x}}$$



Since \hat{x} has fixed length, $\dot{\hat{x}}$ is due to orientation change. $\Rightarrow \frac{d}{dt} \hat{x} = \omega \times \hat{x}$

Introduce cross product matrix $S(a)$, $a \in \mathbb{R}^3$

$$S(a) = \begin{bmatrix} 0 & -a_z & a_y \\ a_z & 0 & -a_x \\ -a_y & a_x & 0 \end{bmatrix}, \text{ note } S^T(a) = -S(a)$$

(3x3)

Now $\dot{\hat{x}} = S(\omega) \hat{x}$.

Also $\dot{\hat{y}} = S(\omega) \hat{y}$, $\dot{\hat{z}} = S(\omega) \hat{z}$.

$$\Rightarrow \dot{R} = \begin{bmatrix} \dot{\hat{x}} & \dot{\hat{y}} & \dot{\hat{z}} \end{bmatrix} = \begin{bmatrix} S(\omega) \hat{x} & S(\omega) \hat{y} & S(\omega) \hat{z} \end{bmatrix}$$

(3x3)

$$\Rightarrow \boxed{\dot{R}(t) = S(\omega) R(t)}$$

Toolbox

$$\text{skew}(\omega) \Rightarrow S(\omega)$$

$$\text{vex}(S(\omega)) \Rightarrow \omega$$

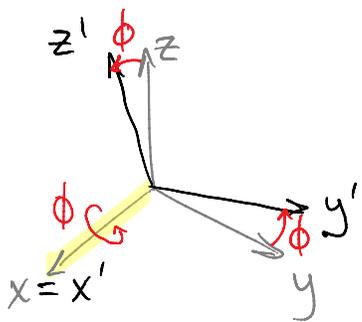
Next, a few specific rate relationships

(not in Corke's text)

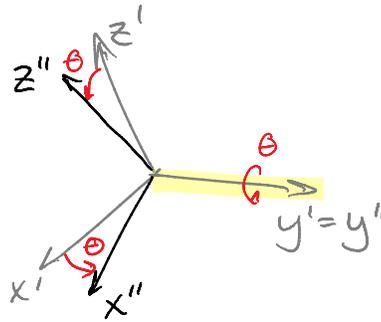
Consider roll, pitch, yaw angles

$$R(\text{rpy}) = R_x(\theta_r) R_y(\theta_p) R_z(\theta_y)$$

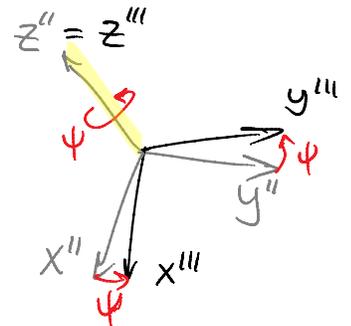
to simplify notation use ϕ, θ, ψ for $\theta_r, \theta_p, \theta_y$



Rotate about
x of base frame



Rotate
about y'



Rotate
about z''

Now suppose we rotate the final frame with contributions $\dot{\phi}, \dot{\theta},$ and $\dot{\psi}$ about each of the highlighted axes?

What would be the angular velocity of the final frame expressed in the base frame?

$$\omega = \mathbb{I} \begin{bmatrix} \dot{\phi} \\ 0 \\ 0 \end{bmatrix} + R_x(\phi) \begin{bmatrix} 0 \\ \dot{\theta} \\ 0 \end{bmatrix} + R_x(\phi) R_y(\theta) \begin{bmatrix} 0 \\ 0 \\ \dot{\psi} \end{bmatrix}$$

$$\begin{bmatrix} \omega_x \\ \omega_y \\ \omega_z \end{bmatrix} = \underbrace{\begin{bmatrix} 1 & 0 & s_\theta \\ 0 & c_\phi & -s_\phi c_\theta \\ 0 & s_\phi & c_\phi c_\theta \end{bmatrix}}_B \begin{bmatrix} \dot{\phi} \\ \dot{\theta} \\ \dot{\psi} \end{bmatrix}$$

$$\boxed{\omega = B\dot{\alpha}}, \text{ where } \dot{\alpha} = [\dot{\phi} \ \dot{\theta} \ \dot{\psi}]^T$$

What about the inverse relationship? Does B^{-1} exist?

Matlab's symbolic toolbox yields:

$$B^{-1} = \begin{bmatrix} 1 & s_\phi \tan(\theta) & -c_\phi \tan(\theta) \\ 0 & c_\phi & s_\phi \\ 0 & -s_\phi/c_\theta & c_\phi/c_\theta \end{bmatrix}$$

Singularity occurs when $\cos(\theta) = 0$

Now consider Euler angle representations.

The exact same approach can be used. The Jacobian will have some singularities!

What about quaternions?

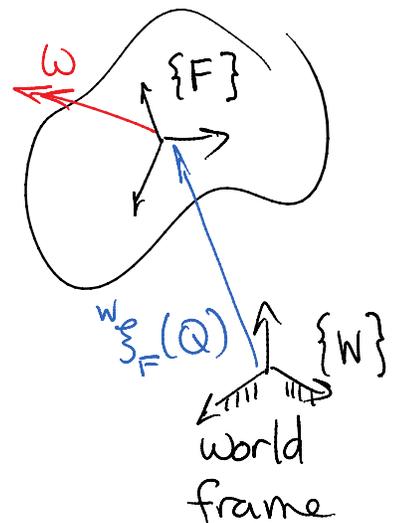
Let us represent \hat{q} as a vector of length 4.

$$\dot{q} \sim Q = \begin{bmatrix} a \\ b \\ c \\ d \end{bmatrix} \begin{array}{l} \text{scalar part} \\ \text{vector part} \end{array}$$

$$\frac{d}{dt} Q = \frac{1}{2} B(Q) \dot{Q}$$

It has been shown that $\frac{d}{dt} Q = \frac{1}{2} B(Q) {}^F \omega$

$$\text{where } B = \begin{bmatrix} -b & -c & -d \\ a & -d & c \\ d & a & -b \\ -c & b & a \end{bmatrix}$$



and ${}^F \omega$ is expressed in the body-fixed frame $\{F\}$

Incremental Relationships

$$\dot{R} \approx \frac{R(t+\delta t) - R(t)}{\delta t}$$

$$\Rightarrow R(t+\delta t) \approx R(t) + \delta t \dot{R}(t) = R(t) + \delta t S(\omega) R(t)$$

$$\Rightarrow \boxed{R(t+\delta t) \approx (\delta t S(\omega) + I) R(t)}$$

$$\delta t S(\omega) + \mathbf{I} = \begin{bmatrix} 1 & -\delta t \omega_z & \delta t \omega_y \\ \delta t \omega_z & 1 & -\delta t \omega_x \\ -\delta t \omega_y & \delta t \omega_x & 1 \end{bmatrix}$$

Note: Finite rotations do not commute, but infinitesimal rotations do! Composing infinitesimal rotations is equivalent to vector addition.

Compose two incremental rotations:

$$R(t + \delta t_1 + \delta t_2) = (\delta t_2 S(\omega_2) + \mathbf{I}) (\delta t_1 S(\omega_1) + \mathbf{I}) R(t)$$

$$= \begin{bmatrix} 1 & -a_z & a_y \\ a_z & 1 & -a_x \\ -a_y & a_x & 1 \end{bmatrix} \begin{bmatrix} 1 & -b_z & b_y \\ b_z & 1 & -b_x \\ -b_y & b_x & 1 \end{bmatrix} R(t)$$

$$= \begin{bmatrix} 1 - a_z b_y - a_y b_z & \vdots & \vdots & \vdots \\ a_z + b_z + a_x b_y & \vdots & \vdots & \vdots \\ -(a_y + b_y) + a_x b_z & \vdots & \vdots & \vdots \end{bmatrix} \begin{matrix} S \\ M \\ L \\ R \end{matrix} \begin{matrix} \vdots \\ \vdots \\ \vdots \\ \vdots \end{matrix} \begin{matrix} S \\ M \\ L \\ R \end{matrix}$$

Assume: $a_i b_j \ll a_k$, $a_i b_j \ll b_k$, $i, j, k \in \{x, y, z\}$

$$\text{Then } R(t + \delta t_1 + \delta t_2) \approx \begin{bmatrix} 1 & -(a_z + b_z) & a_y + b_y \\ a_z + b_z & 1 & -(a_x + b_x) \\ -(a_y + b_y) & a_x + b_x & 1 \end{bmatrix}$$

Note that the result is independent of
the order of incremental rotations

Mapping to incremental Cardan angles

$$R_1 = (\delta t S(\omega) + I) R_0$$

$$\Rightarrow R_1 R_0^T = (\delta t S(\omega) + I)$$

$$\Rightarrow R_1 R_0^T - I = \delta t S(\omega)$$

$$\text{vex}(R_1 R_0^T - I) = \text{vex}(\delta t S(\omega))$$

$$\delta \theta = \delta t \omega$$

$$\delta t \omega \Rightarrow \text{rot}_x(\delta t \omega_x) * \text{rot}_y(\delta t \omega_y) * \text{rot}_z(\delta t \omega_z)$$

result is approx. independent of order

$\delta \theta$ are interpreted as increments of Cardan angles,
a.k.a., roll, pitch, yaw angles.

Note: when approximating R_1 by $(St S(\omega) + I)R_0$,
the result is NOT exactly $\in SO(3)$!
You should normalize. (see 3.2.3)

3.2.2. Incremental Motion (Translation & Rotation)

Now include translation, t

Let $S = \Delta(T_0, T_1)$, where T_1 is very close to T_0

$$S = \begin{bmatrix} t_1 - t_2 \\ \text{vex}(R_1 R_0^T - I) \end{bmatrix}$$

Toolbox: $S = \text{tr2delta}(T_0, T_1)$ $\Delta = \text{delta2tr}(S)$

3.2.3 Inertial Navigation Systems

Orientation of a satellite, robot, etc can be estimated

over time by integrating the output of a inertial measurement unit.

$$R(t+\delta t) = (\delta t S(\omega) + I) R(t) \quad \leftarrow \text{Simplest update scheme. Runge-Kutta and other more accurate schemes are used.}$$

δt is sampling rate
 ω is output of sensor.

Make δt as small as possible to limit error,
i.e. $R(t+\delta t) \notin SO(3)$

One Approach to Normalization of $R(t+\delta t)$

$$\text{Let } R(t+\delta t) = [c'_1 \ c'_2 \ c'_3], \quad R(t) = [c_1 \ c_2 \ c_3]$$

Given ω , one can determine which c'_i
changed the least by:

$$\omega^T R(t) = [d_1 \ d_2 \ d_3]$$

If d_i is $\min(d_1, d_2, d_3)$, then c_i changed least.

For argument sake, assume c_3 changed least.

$$\text{Let new } c_3'' = c_3' / \|c_3'\|$$

$$c_1'' = (c_2' \times c_3'') / (c_2' \times c_3')$$

$$c_2'' = c_3'' \times c_1''$$

Corke normalizes
all vectors last.

Using quaternions to increment orientation.

$$\frac{d}{dt} \dot{q} = \frac{1}{2} \dot{q}(\omega) \dot{q} \leftarrow qd = q \cdot \text{dot}(\overbrace{[1, 2, 3]}^{\omega})$$

$$\dot{q}(t+st) = \dot{q}(t) + st \frac{d}{dt} \dot{q}(t) \leftarrow q2 = q1 + 0.001 * qd$$

$$\dot{q}(t+st)' = \dot{q}(t+st) / \|\dot{q}(t+st)\| \leftarrow q2 = q2 \cdot \text{unit}()$$

The matrix-vector implementation of $q \cdot \text{dot}(\omega)$ was given above. Recall:

$$\frac{d}{dt}(\dot{q}) = \frac{1}{2} B(\dot{q})^F \omega$$

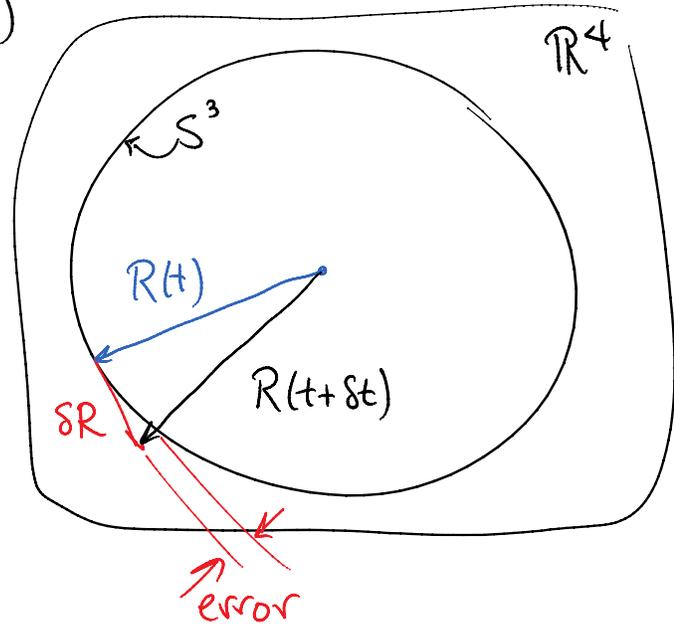
where $\{F\}$ is the frame fixed in the rotating body.

One last point about incremental changes to orientation.
 $SO(3)$ is a sphere, S^3 , embedded in \mathbb{R}^4 .

The simplest update (used above)
is a vector tangent to S^3
added to a vector in S^3 .

The result is not in S^3 .

\therefore Normalize!



More sophisticated updates

such as Runge-Kutta updates tip δR so that it
becomes a chord.

